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EXPLOSIVE-WELDED FIBER REINFORCED MATERIALS:

STATUS OF TECHNIQUE AND POSSIBILITIES OF DEVELOPMENT

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16. Abstract

A wide variety of fibers (boron, V2A, tungsten, molybdenum, etc.) have been successfully imbedded in a number of metal matrices (aluminum, nickel, zirconium, titanium) using an explosive welding technique. Semi-finished products can also be manufactured to provide shapes such as wing profiles, blades, etc. Tensile strengths measured have been as high as those above theoretically predicted. Mass production methods are also discussed. The materials are extremely heat resistant and it is thinkable that 150 cm fiber reinforced turbine blades could be made which could withstand temperatures up to 1100° C.

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EXPLOSIVE-WELDED FIBER REINFORCED MATERIALS:
POSITION OF TECHNIQUE AND POSSIBILITIES OF DEVELOPMENT

E. Wolff

1. INTRODUCTION

Explosive welding is one of the methods of producing fiber-reinforced metals which has a special significance because of the universal applicability to almost all fiber/matrix combinations, regardless of their chemical compatibility or the melting points of their components. The first attempts to produce metallic fiber composite materials were carried out in 1968 in England [1]. Fundamental research was done in the United States on composite materials produced in this way [2, 3] and these experiments also started with a matrix in foil or sheet form as the initial material.

Fleck [4], however, attempted to produce explosive imbedding of metallic reinforcement fibers in a powder metal matrix. The strength properties of these composite materials, however, are not at all comparable with those which can be produced from a foil matrix, disregarding the fact that in every case the sintering process which follows and which is absolutely necessary leads to a considerable destruction of the composite material by diffusion between the composite components. Therefore, the

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* Numbers in margin indicate pagination in original foreign text.

method becomes uneconomical and meaningless because of the high degree of complexity.

In 1970 the author carried out experiments for producing fiber-reinforced metals by explosive welding [5]. He used metal foil as the initial material again. A hollow body was prepared using special winding techniques which had fibers in the axis direction. The individual foil layers were welded by the action of explosives. In these investigations we were concerned with the production of profiled hollow bodies which can be produced with relatively small degree of complexity and with good reproducibility because of the further advantage in the technology of semifinished product production. In contrast to this, earlier authors carried out fundamental research in this area.

2. METHODS OF EXPLOSIVE WELDING

The distance between the two metal plates which are to be welded using explosive welding techniques is one characteristic parameter. When the explosive is detonated on one plate, it is accelerated towards the other plate and reaches it with a high velocity. This collision velocity is required in order to introduce processes which occur during the explosive welding process, such as tearing of the surface layer, wave formation, etc [6]. During the production of fiber composite materials by explosive welding, it is also necessary that the metal foils have an exactly defined distance from each other. This distance is prescribed by the fiber diameter, and fortunately by modifying the explosive charge thickness and composition, it is possible to obtain perfect welding results over a large range of fiber diameters and foil thicknesses. This means that almost any desired fiber volume fraction can be realized.
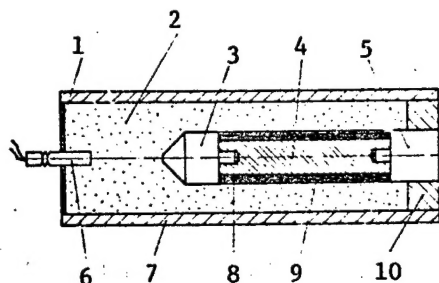
Figure 1. Configuration for
explosive welding of fiber
reinforced hollow bodies.

1- cardboard disk; 2- explosive;
3- head; 4-core; 5- tear away
plug; 6- igniter capsule;
7- plastic tube; 8-covering
tube; 9- foil plus fibers;
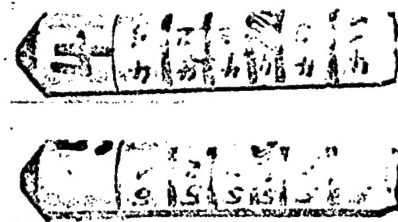10- plastic material disk.

Figure 2. Bombarded fiber
reinforced hollow bodies.

Figure 1 shows the structure of a laboratory configuration
for a hollow cylinder which is fiber-reinforced in the axis
direction. The matrix foil with the fibers attached to it is
wound in several layers around a cylindrical core, and an
alternating configuration of fiber layers and foil layers occurs.
The cross-sectional shape of the core determines the later
cross-sectional form of the fiber-reinforced hollow body, and
has a metallic covering tube which is placed over the winding
body. The winding body is closed at both ends with plugs, so
that the detonating explosive will not tear up the winding. The
body prepared in this way is concentrically secured with a plastic
tube and the slit surrounding it is filled with explosives. It
is ignited on the end while hanging over a sand pit. As a rule
these bodies can again be found after explosion. The temperature
of the parts (Figure 2) ranges from hand temperature up to about
150° C and depends essentially on the fiber volume fraction, i.e.,

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on the plastic work of deformation which the matrix must absorb. The air between the fibers and the foils is pressed downwards for the ring-shaped detonation progressing in the axial body direction. Of course, it also heats up adiabatically or gives part of its considerable heat to the body, so that the body has a heat profile which increases in the direction of detonation. For very high fiber volume fractions (> about 50%) it can become necessary to detonate the parts in a vacuum for certain fiber/matrix combinations.

The removing of the cores and the covering tubes of course represents undesirable additional work, which can be avoided by covering the core with a plastic layer (shrinkable tubing) and instead of the metallic covering tube, one winds a self-adhesive aluminum foil around the body in several layers. The foil then explodes during the detonation and the body is displaced partially or even completely away from the core covered with plastic, which then can be completely removed. However, no additional work is required on the composite body.   /288

3. PROCESSES DURING THE EXPLOSIVE WELDING PROCESS

During the detonation of the explosive, pressures up to 200,000 bar act on the metal foils to be welded to one another. In this way, because of the high deformation velocity, the flowing materials show a behavior which can be described using the laws of hydrodynamics, i.e., the materials can be considered as liquids having greater or smaller viscosity. Figure 3 gives a clear example of this, and shows a V2A fiber welded into a nickel foil. Because of the oblique position with respect to the direction of detonation, the incident matrix metal flow was oblique with respect to its axis and it therefore approximately deformed into a Joukowski profile.

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If the fibers are stretched tightly between the foils and if the detonation wave runs in the axial direction, then if the explosive is correctly selected and for a correspondingly increased ratio of fiber hardness to matrix hardness, one obtains an exceptional welded composite body (Figure 4). The matrix material squeezed between the fibers takes on the profile of a laminar flow. Since the fibers are pressed into the foil below at the same time, the material of this foil flows upward with the same flow profile. The two profiles meet at the height of the fiber center at a point, if we consider a planar model, from which point the welding between the foils progresses in a direction oblique with respect to the fibers (Figure 5).

4. ATTACHMENT OF FIBERS ON THE FOILS

It was shown that a tight stretching of the fibers in the winding body was one prerequisite for the production of a perfect composite body. For this purpose, the fibers must have a certain prestress when they are attached to the foils, and many methods can be used.

One method with a low degree of complexity consists of winding around a foil strip which is stretched tightly between the headstock and and chuck of a winding machine (Figure 6). The beginning and ending of the wire are directed through small holes at the edges of the strip and are knotted there. This foil strip covered with fiber is wound around a prismatic core together with a strip with no fibers, so that an alternate layering of foil and fiber layers is produced. The method, however, has the disadvantage that, if relatively soft matrix metals are used, the foil edges will bend because of the fiber tension, so that the fiber stress becomes variable over the foil strip length when the foil is stretched later on. The method is therefore only suitable for relatively stiff matrix materials and relatively

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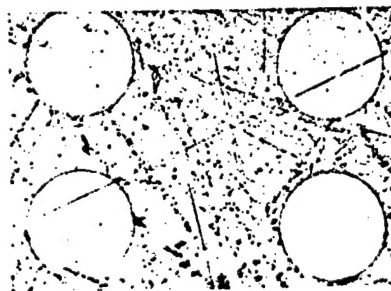Figure 3. Deformation of a V2A fiber in a Ni matrix into approximately a Joukowski profile.

Figure 4. Welded composite material free of defects (fiber: tungsten 0.1 mm; matrix: niobium 0.1 mm).

Figure 5. Welding of matrix foils with collision point located at the center between the fibers (fiber: V2A 0.1 mm; matrix: nickel 0.1 mm)
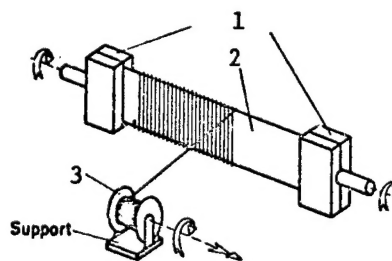

Figure 6. Covering of foil by layers using winding.
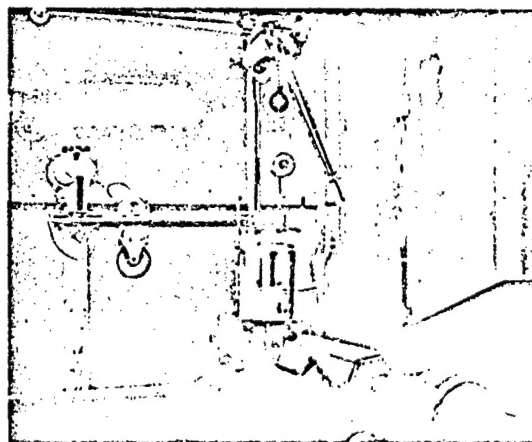
1- clamping heads; 2- sheet strips; 3- fiber coil;

Figure 7. Configuration for winding the auxiliary cylinder with fibers.

small fiber volume fractions.

The second method for attaching the fibers to the foils uses an auxiliary cylinder with a multicorner cross section, and the foil strips are mounted on to its side so that they can be removed. This auxiliary cylinder is now clamped in the same winding machine (lathe) and fibers are wound on it. The fiber volume fraction desired for the fiber composite material is produced, on the one hand, by the fiber diameter to foil thickness ratio and also by the machine advance within wide limits. The prestress with which the fibers are wound on the auxiliary cylinder depends on the materials used as well as on the foil and wire thicknesses. The uniformity of the stress is provided by a special thread stand with a control device for the thread stress (Figure 7).

The following methods have proven themselves for attaching the fibers at the edges of the foil.

4.1. Gluing of the fibers

The fibers are attached to the foil edges with an adhesive layer about 5 mm wide and the metal adhesive can be applied either before or after the winding process by means of a roller (Figure 8). There is no welding between the metal foils at the points with adhesive. However, we were not able to detect any extensive prevention of welding, for example, by vaporized adhesive. The adhesive can only have a detrimental effect if it partially covers the fibers with a thin layer. During rolling up of the foils on the cores, this leads to thickening at the ends of the winding body. Therefore, it is often difficult to maintain the maximum foil separation specified by the fiber diameter. Since the experiment was always adjusted to a foil separation corresponding to the fiber diameter, a greater separation leads to melting
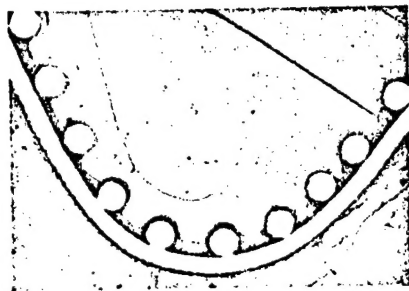
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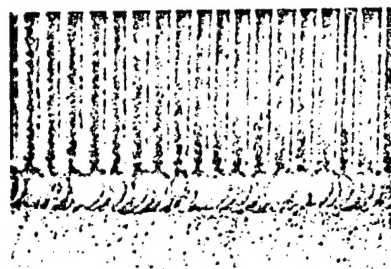Figure 8. Fibers glued on to foil in a cross section (fiber: V2A 0.1 mm; foil: nickel: 0.05 mm).

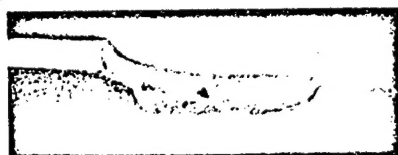Figure 9. V2A fibers (0.1 mm) welded to a nickel foil (0.1 mm) using an electron beam.

Figure 10. Longitudinal section through the electron beam welding of Figure 9.

Figure 11. Roller welding of molybdenum fibers (0.1 mm) on nickel foil (0.05 mm).

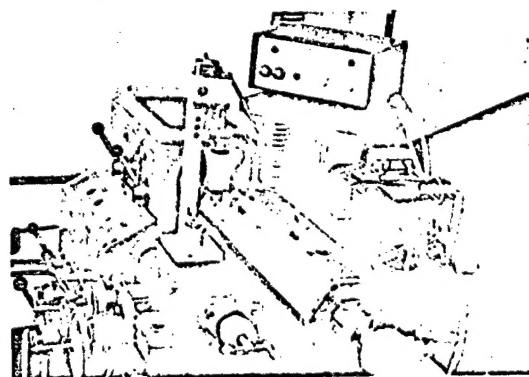Figure 12. Longitudinal section through roller welding of Figure 11.

Figure 13. Configuration for roller seam welding of the fiber/foil semifinished product.

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because of the then considerably increased contact velocity between the foils, which can make the composite material useless.

4.2. Welding of fibers with the electron beam

Figures 9 and 10 show an example of an electron beam weld between stainless steel fibers and a nickel foil (or welding of stainless steel fibers to a titanium foil). This shows that exceptional results can be achieved with this process. The great complexity of this method means that the semifinished products cannot be produced using this method economically in the near future.

One positive feature of electron beam welding is the fact that it is possible to weld a wide range of fiber and matrix materials with different diameters and thicknesses, without changing the machine adjustment substantially. In no case did we observe drop formation at the welding points so that when the winding bodies are produced, we no longer have the disturbing thickened parts along the face sides of this body. Strength measurements at the welding points in the fiber direction showed that the fibers can accept a multiple of the required prestress force, even though in general brittle intermetallic compounds are formed in the welding zone.

4.3. Welding of fibers with the welding roller

The method showed exceptional results just like the electron beam welding method. However, it has the distinct advantage that it can be automated to a high degree and can be implemented with a low degree of complexity. Such welds between molybdenum fibers and nickel foil is shown in Figure 11. Figure 12 shows a longitudinal section through the welding point for this combination of materials. Here again the strength of the weld in the fiber
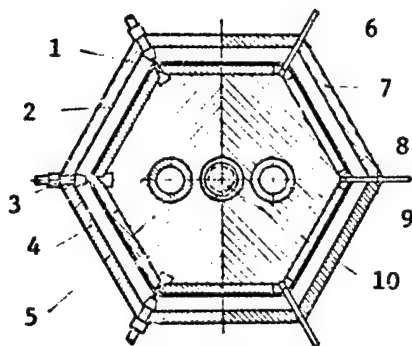
Figure 14. Possibilities for separating the fiber winding at the edges of the auxiliary cylinder.

1- fiber winding; 2- foil;
3- oxyhydrogen gas burner;
4- copper plate; 5- core;
6- knife roller; 7- frame;
8- knife bar; 9- guide jack;
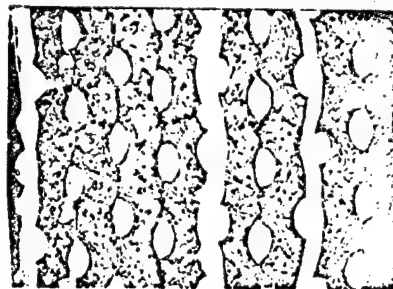10- transport spindle.

Figure 15. V2A tissue (wire thickness 0.1 mm) reinforced nickel matrix (0.1 mm).

direction is a multiple of the strength required for absorbing the fiber prestress, and there is a small amount of scatter in the strength values.

A transformer, a control unit and a spring-suspended welding roller are required for the process. The welding head was attached to the support of a lathe (Figure 13) during the experiment and it was possible to use its automatic advance so as to regulate the welding velocity, which is on the order of 10 mm/sec. The current supply through the auxiliary cylinder made of copper plates was through both face sides of the cylinder in order to reduce the voltage drop during the welding process.

5.   FURTHER PROCESSING OF THE SEMIFINISHED PRODUCT

The following discussion is limited to the semifinished product obtained using the winding method with the auxiliary cylinder and by welding the fibers with the welding roller, because these methods have special importance compared with the other methods discussed before.

In order to remove the foils attached to the auxiliary cylinder and the welded fibers, it is necessary to separate the fibers at the edges of the auxiliary cylinder. For this purpose, the cylinder has longitudinal grooves at the edges, so that two technical-economical methods can be used for separating the fibers (Figure 14).

The natural method is to cut through or shear off fibers at the sharp and hard edges of the grooves with a knife roller. The second method consists of burning through the fibers using a oxyhydrogen gas burner, which can be used to easily separate the fiber materials with a high melting point, such as, for example, tungsten. In both methods it is possible to separate the fibers at all edges of the auxiliary cylinders, by arranging the separation devices in a special device, in which the cylinders are transported by means of threaded spindles or similar devices.

One can easily imagine a combination of the welding and separation processes for automatic production, if the welding rollers are arranged ahead of the separation elements in such a device. However, a separate current supply would have to be provided for each welding roller in order to provide for uniform supply of all the welding rollers with a current of equal intensity. Immediately after welding the fibers to the foils, it would be possible to separate the fibers by the subsequent knife-roller or burner along the edges of the auxiliary cylinder.

This would mean that the foil strips covered with the fibers could then be removed for further processing from the auxiliary cylinder.

The further processing of the semifinished product is done by winding a core with this foil or by applying layers, depending on whether we have a hollowed body or a flat body. In both cases one must make sure that a very dense packing is brought about.

When producing flat fiber reinforced plates, one side of the layer is covered with explosives after it has been covered, for example, by self-adhesive metal foils and is therefore protected against combustion products of the detonating explosive. Such a layering requires a small degree of effort when using metallic tissue. In this way, a body is produced which is reinforced in two directions (Figure 15) and its tensile strength in the fiber direction is considerably lower than for a body reinforced in a unidirectional direction with a comparable fiber volume fraction, because of the crossing points with the fibers which run transversely.

The purpose of this work was to produce fiber-reinforced hollow bodies with a profiled cross section. We proceeded by winding the core, corresponding to the cross-sectional profile being produced, with the foil covered with the fibers. The fibers run in the axial direction which at the same time is also the direction of detonation. It has been found that it is useful to apply a plastic layer (shrinking tubing) to the core so that, after bombardment, it can more easily be removed from the hollow body. The end of the winding is prevented from coming apart by means of an adhesive foil and the winding body is surrounded with a metallic covering, in order to protect it against detonation products. Instead of the massive metal covering, it is also possible to wind on a self-adhesive metal form in several layers on to the body. It explodes during
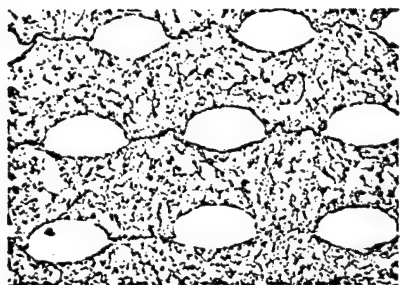
12

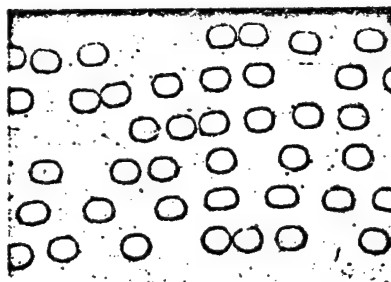Figure 16. Explosively welded composite material of Ni foil (0.1 mm) with V2A fibers (0.05 mm).

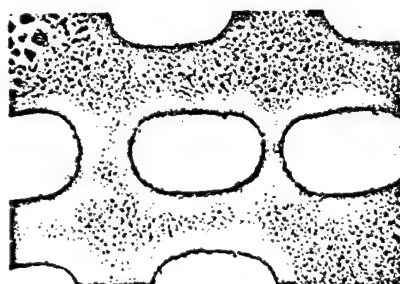Figure 17. Explosively welded composite of V2A foil (0.1 mm) with tungsten fibers (0.1 mm).

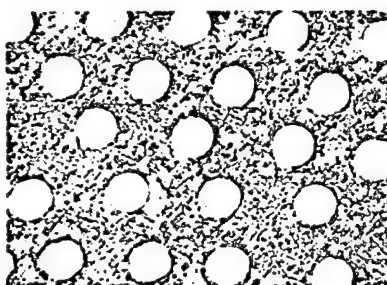Figure 18. Explosively welded composite of spring steel foil (0.09 mm) with tungsten fibers (0.1 mm).

Figure 19. Explosively welded composite of titanium foil (0.1 mm) with tungsten fibers (0.1 mm).
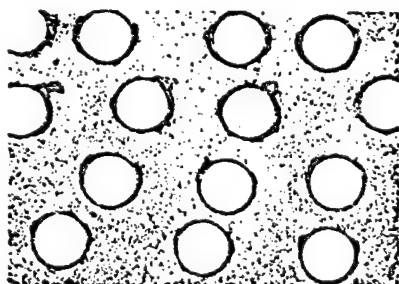
Figure 20. Explosively welded composite of zirconium foil (0.1 mm) with tungsten fibers (0.1 mm).
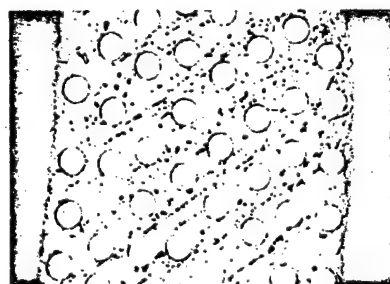
Figure 21. Explosively welded composite of niobium foil (0.1 mm) with tungsten fibers (0.1 mm).

detonation, so that it is no longer necessary to dig the core
and the surrounding tube out after detonation. The winding body
is closed off along both faces with metal or plastic plugs, so
that no explosive can reach the layers to be welded. The
laboratory configuration is the one described in Section 2.

6. FIBER COMPOSITE MATERIALS WHICH CAN BE PRODUCED BY EXPLOSIVE WELDING

Almost all metallic fiber/matrix combinations can be produced
using the explosive welding technique, if these materials have a
certain minimum amount of ductility. Because of the nature of
fiber reinforcement, the fibers are usually much harder than the
matrix materials used, so that fiber deformation must usually be
limited. The E modulus ratios, the heat expansion ratios and the
chemical compatibility of the components do not have a direct
influence on the effectiveness of the explosive welding process.
They only become important according to the applications, for
example, if a high temperature load is required of the composite
material.

Figures 16 to 22 show cross sections of successfully welded
composite materials. The limits for the maximum possible fiber
volume fractions are not specified by the method, but by the
geometric conditions and the minimum distances between fibers
determined for physical reasons, which can be assumed to lie
between 5 and 10 µm. Figure 23 also shows that the matrix
material can flow into very narrow cracks when the explosive
welding technique is used. The smallest crack filled by
the titanium matrix is about 3µm here.

The imbedding of brittle fiber metals such as, for example,
berylium (Figure 24) causes difficulties, which usually fractures
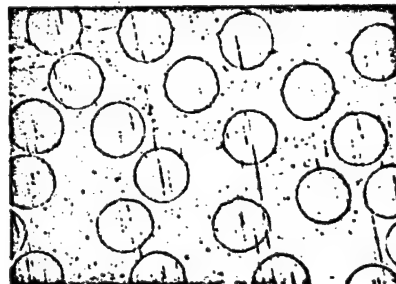even if soft matrix metals are used such as aluminum.

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Figure 22. Explosively welded
composite of inconel-718 foil
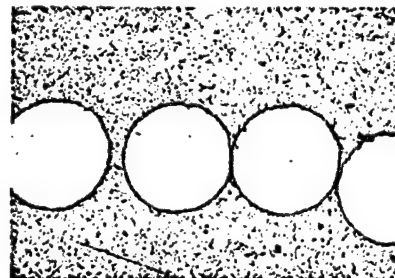(0.053 mm) with tungsten fibers
(0.1 mm).

Figure 23. The matrix metal
flows in even the smallest
spaces between fibers during
explosive welding.

Figure 24. Explosively imbedded
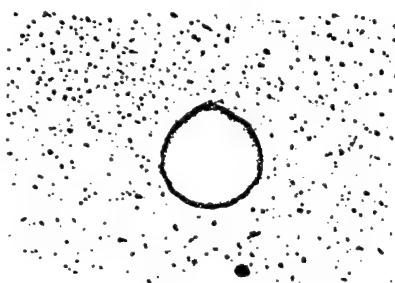berylium fiber (0.1 mm) between
nickel foils.

Figure 25. Explosively imbedded
ceramic fiber (Tyco 0.3 mm) be-
tween niobium sheets.

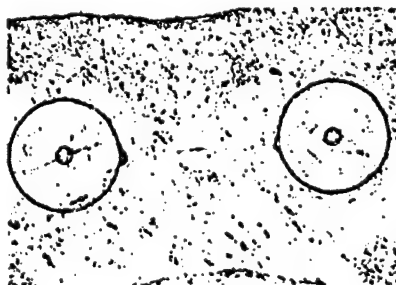Figure 26. Explosively imbedded
boron fibers (0.1 mm) between
nickel foils.

Figure 27. Explosively imbedded
carbon fibers (5 μm) between
nickel foil.

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The imbedding of ceramic fibers (Figure 25), boron fibers
(Figure 26) and carbon fibers (Figure 27) without destruction is
even more difficult and sometimes impossible. As a rule, these
fibers break up into short pieces unless they are completely
compressed.

7. PROPERTIES OF EXPLOSIVELY WELDED FIBER COMPOSITE MATERIALS

There is a true welding between the matrix foils during the
explosive welding of fiber composite materials. One condition for
this is that the foils have a separation which makes it possible
to accelerate them to the required collision velocity. At the
collision point a beam of matter is formed which consists of
particles of both foil surfaces, and which flies away in the
direction of detonation [6]. The metal surfaces which have become
blank weld together under the effects of the detonation pressure.
At the present time we were not yet able to prove to what extent
this is also true for the boundary between the fiber and the
foils. However, it is certain that even the microscopically
smallest surface roughnesses are filled with matrix metal (see
Figure 23), so that an extremely intensive mechanical clamping
between the fibers and the matrix is brought about, even though
this is usually not a true metallurgical bond. However, a
diffusion barrier remains intact which later on has a positive
effect for high temperature operation.

The appearance of the fracture surfaces gives an indication
on the quality of the bond. If the fibers tear at the fracture
surface of the matrix without standing out (Figure 28), we can
conclude that we have a very good bond.

The tensile strength in the fiber direction measured on
the bodies shown here usually amounted to between 80 and 100% of
the tensile strength determined according to the mixing rule.

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Figure 28. Longitudinal section through a tearing sample made of Inconel 718 with tungsten fibers (0.1 mm).
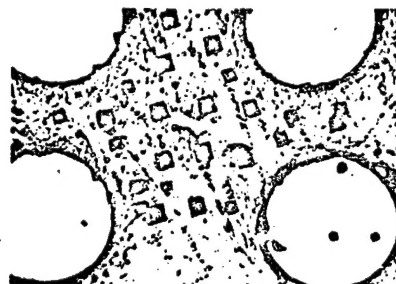

Figure 29. Strengthening of a titanium matrix by plastic deformation during explosive welding.

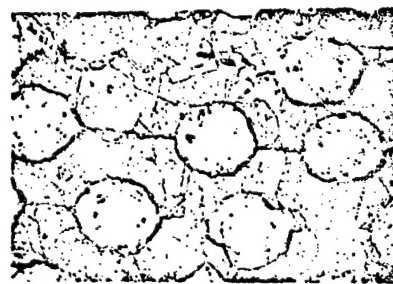Figure 30. Nickel matrix with V2A fibers, annealed for 9 hrs. at 800° C.

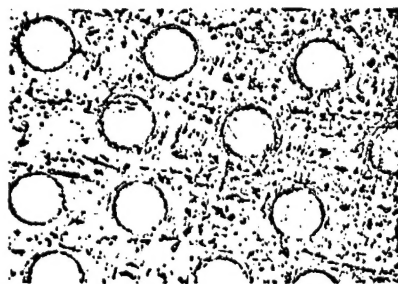Figure 31. Nickel matrix with V2A fibers, annealed 96 hrs. at 800° C.

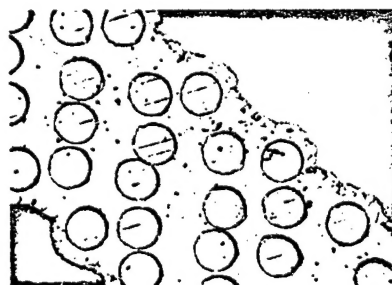Figure 32. Niobium matrix with tungsten fibers, annealed for 96 hrs. at 1100° C.

Figure 33. Inconel-718 matrix with tungsten fibers, annealed 96 hrs. at 1100° C.

For the composite material shown in Figure 28, tungsten fibers
and an Inconel-718 matrix with a fiber volume fraction of 28%,
we measured a tensile strength of 101 kp/mm^2, for example, on the
average, and the calculated tensile strength is 116 kp/mm^2.

When the fibers are surrounded with a material, the matrix
is deformed in a cold state, which in some matrix materials
brings about a strong strengthening (Figure 29). In this way
the strength of the matrix is increased and often a higher tensile
strength is determined than was predicted by the calculation.
Explosive welding as a method for producing fiber composite
materials is suitable most of all for composite materials with
a high degree of temperature resistance. Therefore, in particular
we are interested in the compatibility of the components at
temperatures to which the materials could be subjected depending
on the applications. Figures 30 to 33 show a few results of
annealing experiments with various material combinations. In
the case of a combination of V2A fibers and a nickel matrix, we
find after a 96-hour annealing at 800° C that there is a strong
diffusion between the fibers and the matrix (Figure 31); the
composite has already been extensively damaged. A niobium matrix
with tungsten fibers after 96 hours of annealing at 1100° C
(Figure 32) also reacts strongly. The technically very interest-
ing combination of tungsten/inconel-718 after 96 hours of
annealing at 1100° C shows only a very narrow diffusion zone
(Figure 33) even though these fibers do not have any alloy
additives which would lead one to expect a considerable improve-
ment in the compatibility.

8. FORMING POSSIBILITIES AND APPLICATIONS

Hollow bodies of widely varying cross sections can be /293
produced using the winding techniques discussed. If the core
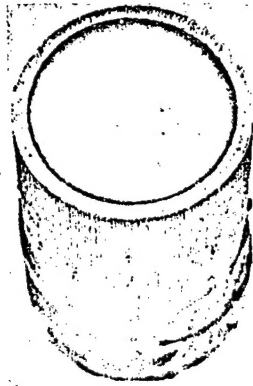has the proper shape, it is not necessary to post-form the

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Figure 34. Explosively welded
fiber reinforced hollow cylin-
der (fiber: tungsten 0.1 mm;
matrix: zirconium 0.1 mm;
fiber volume fraction 24%).

Figure 35. Explosively welded
fiber reinforced hollow body with
blade cross section. (fiber:
tungsten 0.1 mm; matrix: inconel
718 0.08 mm).

cross-sectional shapes, so that hollow bodies produced in this
way practically do not have to be machined. Figure 34 shows a
fiber-reinforced hollow cylinder with a diameter of 17 mm and a
length of 60 mm produced according to this method.

The hollow body with the blade cross section shown in
Figure 35 was also produced using this method and no additional
complexity was involved. Blade profiles with a concave under-
side are also possible. By further refinement of this technique
one could certainly imagine that it would be possible to weld in
ribs into the hollow body during the welding of the foil winding.
On the other hand, the installation and forming of the blade foot
with additional stiffening elements will require one of several
more working processes. From the point of view of the technology
of today, it will be possible to produce fiber reinforced tur-
bine blades up to a length of about 150 mm using the explosive
welding technique, and even relatively flat profiles can be pro-
duced. If it becomes possible to improve the compatibility of
the fiber and matrix materials further, and to develop usable

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protective layers for blades against corrosion and erosion,
then explosive welding can contribute to the building of turbine
blades which can withstand temperatures of 1100° C and more, such
as are required at the high rpms of today.

REFERENCES

1. Jarvis, C. V. and P.M.B. Slate. Explosive Fabrication of
   Composite Materials. Nature, Vol. 220, No. 5169, 23. 11.
   68, pp. 782-783.

2. Fleck, J. N., D. Laber and R. W. Leonard. Explosive
   Welding of Composite Materials. J. of Composite Materials,
   Vol. 3, October 69, pp. 699 - 701.

3. Reece, O. Y. Explosive Bonding Packs Strength into Metal
   Composites. Iron Age, Vol. 29, 29. 1.70, pp. 60-61.

4. Fleck, J. N. Non-conventional Fabrication of Metal-Matrix
   Composites. ASME Internat. Engineering Conf. and Tool
   Exposition, Detroit, 13.-17.4.70, Paper EM 70-125.

5. Wolff, E. Production of Thermally and Mechanically
   Highly Loaded Components, especially Turbomachine Blades
   made of Fiber-Reinforced Metals, using the Explosion
   Forming Technique. German Patent Application,
   No. 2055637, 12. 11.70.

6. Crossland, B. The Development of Explosive Welding and
   Its Applications in Engineering. Metals and Materials,
   Dec. 71, pp. 401-413.